Dual-Functional UAV-Empowered Space-Air-Ground Networks: Joint Communication and Sensing

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Abstract—In this paper, we investigate a sensing-enabled integrated space-air-ground (SAG) data collection network, in which an unmanned aerial vehicle (UAV) can not only work singly to sense data from multiple targets but also collaborate with a lowearth orbit (LEO) satellite to collect communication data from multiple users. Since the coverage of the UAV is much smaller than that of the LEO satellite, we first determine the set of usable users and targets for the UAV by analyzing the signal-to-noise ratios between the UAV and the users and targets. Based on this, we pose an optimization problem designed to maximize the total amount of data collected in the network while satisfying the constraints of UAV energy consumption, memory capacity, and minimum amount of sensor data per target. Moreover, considering that the network consists of three layers and the UAV has dual functions of communication and sensing, this problem is solved by jointly optimizing the scheduling of the users' data upload scheme, the UAV trajectory, and the allocation of communication and sensing time. However, the formulated problem is a mixed integer nonlinear programming (MINLP) problem, so it is difficult to find the optimal solution. Therefore, we further design an alternating iterative optimization algorithm (AIOA) framework to find an appropriate solution. Specifically, we alternately optimize the UAV trajectory, time allocation strategy, and data upload schedule in each iteration. Finally, simulation experiments validate the effectiveness of the AIOA and its superiority over other benchmarks in terms of the amount of data collected.

Index Terms—SAG network, UAV trajectory optimization, data collection, communication and sensing.

I. INTRODUCTION

As society accelerates its transformation to digitalization and intelligence, sixth-generation (6G) technology is developing rapidly. 6G can provide ultra-high speed, low-latency, and high-reliability communications to meet the requirements of future 6G applications [1], [2]. Meanwhile, 6G technology is also facilitating the development of Internet of Things

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(IoT) networks, leading to numerous IoT applications such as smart environmental monitoring, efficient energy management, and precision agriculture technologies [3]–[6]. However, most 6G-enabled IoT applications rely heavily on efficient and reliable data collection and processing capabilities, making data collection and analysis particularly critical. Although 6G networks can offer unprecedented communication capabilities, this challenge still exists in remote and topographically complex areas. To meet this requirement, an integrated Space-Air-Ground (SAG) network has been proposed as a promising solution for data collection, with the unique advantage of being able to integrate satellite and airborne platform systems [7]–[9]. In particular, in remote areas that are difficult to cover with traditional communication technologies, the SAG network shows great potential [10].

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Specifically, the integrated SAG network includes satellites, airborne platforms and ground nodes, with satellites and airborne platforms playing a central role in the SAG network structure. On the one hand, satellite communications are provided by geostationary orbit (GEO), medium earth orbit (MEO) and low earth orbit (LEO) satellites, each of which has its own specific service characteristics and application scenarios [11], [12]. In particular, GEO satellites are mainly used for broadcasting and weather monitoring, and MEO satellites are suitable for navigation services. LEO satellites are particularly suitable for the realization of SAG networks due to their low latency and high speed data transmission caused by their low orbits [13]–[15]. In this field, The StarLink project, recently built by SpaceX, has been proposed to build a constellation of a large number of small satellites in LEO to provide high-speed broadband Internet services to users around the world. In addition, the authors in [16] investigated downlink transmission strategies for large-scale multiple-input multipleoutput (MIMO) LEO satellite communications systems. In addition, the authors in [17] investigated the reliable performance of downlink LEO satellite communication systems and also investigated UAV-based cooperative transmission to compensate for the large path loss caused by long transmission distances.

While LEO satellites provide extensive coverage for global connectivity, they still have limitations, such as high latency compared to terrestrial communications. In response, UAVs are being used as carriers for information acquisition, transmission and processing, complementing and extending LEO satellite communications. Taking advantage of UAV characteristics such as lower latency, high mobility, rapid deployment, and line-of-sight (LoS) channels, UAVs can serve as relays to effectively overcome the limitations of LEO satellites and improve the overall performance of the SAG network. In particular, UAVs can be rapidly deployed in targeted areas to provide various transmission services, such as point-to-point and wide-area data transmission services. When providing services, UAVs can operate in two communication modes: "hovering communication" and "flying communication", depending on whether the UAV is hovering or flying when providing services. Although UAVs have such advantages and benefit the SAG network, their limited battery capacity and high energy consumption should be considered to provide more sustainable services. To this end, the tradeoff between UAV energy consumption and information throughput has been extensively studied in the literature, such as [18], [19]. For example, the researchers in [19] aimed to minimize the total energy consumption of the UAV while satisfying the communication throughput requirements of the ground nodes. Therefore, they formulated and addressed an optimization problem involving complex considerations of UAV trajectory, communication time allocation, and mission completion time.

In the recent vision of wireless communications, 6G will not only provide ultra-high-speed, low-latency, high-reliability communication services, but also extend to high-precision sensing services [20]. However, as an important component of 6G, the SAG network has seen less research in the area of sensing, although there have been some studies exploring the use of UAVs in sensing missions [21]-[23]. In particular, a base station with dual-function radar communication (DFRC) capability can not only provide communication services, but also sense the environment, e.g. by collecting information on the target's position, speed, and shape, as well as information on human posture from radar echoes [20], [24]. However, due to limited resources, how to design DFRC schemes to balance communication and sensing performance is still a challenge. In this area, the UAV was designed to alternately transmit communication signals and sensing signals to achieve DFRC [25], [26]. In addition, the base station could transmit communication and sensing signals simultaneously to realize DFRC [27], [28]. In addition, a dual-function beamforming matrix of communication and sensing was designed to achieve DFRC [29]. Although the difference exists in the implementation of the aforementioned DFRC methods, the common goal of those is to optimize the efficiency of communication while ensuring the quality of sensing. Motivated by this, we consider a novel integrated SAG network, where the UAV can provide both communication and sensing services during flight.

In this paper, we study a sensing-enabled integrated SAG data collection network, where the UAV can not only work singly to sense data from multiple targets but also collaborate with the LEO satellite to collect communication data from multiple users. By taking into account that the service coverage of the UAV is much smaller than that of the LEO satellite, we first determine the set of serviceable users and targets for the UAV based on the signal-to-noise ratios (SNRs) between the UAV and users and targets. From this, we pose an optimization problem designed to maximize the total amount of data

collected in the network, while simultaneously meet the constraints of the UAV energy consumption, memory capacity, and minimum amount of sensing data per target. Moreover, considering that the network encompasses three layers and the UAV has dual functions of communication and sensing, this problem can be solved by jointly optimizing the scheduling of the users' data uploading scheme, UAV trajectory, and allocation of communication and sensing time. However, the problem is a mixed integer nonlinear programming (MINLP) problem that is difficult to solve. To deal with the problem, we propose a two-stage alternating iterative optimization algorithm (AIOA) framework to find a proper solution. Specifically, in the first stage, we employ the successive convex approximation (SCA) algorithm to optimize the UAV trajectory. In the second stage, with fixed UAV trajectory, the block coordinate descent (BCD) technique is utilized to obtain a time allocation strategy and users' data upload scheme. Eventually, simulation experiments validate the effectiveness of the AIOA and its superiority compared to other benchmarks regarding the amount of data collected. The main contributions of this paper are as follows:

- We investigate a sensing-enabled integrated SAG data collection network, in which the UAV can not only work singly to sense data from multiple targets but also collaborate with the LEO satellite to collect communication data from multiple users.
- We design an optimization problem to maximize the total amount of data collected in the network, while simultaneously meet the constraints of UAV. We solve this problem by jointly optimizing the scheduling of the users' data uploading scheme, UAV trajectory, and allocation of communication and sensing time.
- We propose an effective two-stage AIOA framework to find a proper solution to the formulated problem. Specifically, in the first stage, we employ the SCA algorithm to optimize the UAV trajectory. Once the UAV trajectory is determined, the BCD technique is utilized in the second stage to obtain a strategy of time allocation and scheduling of users' data upload scheme.
- We conduct extensive simulation experiments validate the effectiveness of the AIOA and its superiority compared to other benchmarks regarding the amount of data collected.

II. SYSTEM MODEL

In this section, we first introduce the network model of SAG networks, and then describe the data collection in multi-user multi-target scenarios. After that, we detail the data collections of communication and sensing.

A. Network Model

As shown in Fig. 1, we consider an integrated SAG data collection network, where the UAV and the LEO satellite are used to cooperatively collect data from N communication users (CUs) and M sensing targets. Let $C \triangleq \{c_1, \dots, c_n, \dots, c_N\}$ and $S \triangleq \{s_1, \dots, s_m, \dots, s_M\}$ denote the sets of CUs and sensing targets, respectively. In this network, the memory and energy limited UAV is sent to collect data from user c_n and target s_m according to a designed trajectory. Specifically, two



Fig. 1. Illustration of SAG networks for the collection of communication data and sensing data.

types of data need to be collected, namely, communication data and sensing data. For the communication data collection, the UAV serves as an air base station or relay to assist the LEO satellite in data collection, i.e. c_n can upload its data directly to the LEO satellite or indirectly through the UAV. For sensing data collection, the UAV serves as a radar to sense data from the target s_m , such as information about the target's position, velocity, and shape, as well as information about the target's attitude. Although the UAV shows great flexibility in data collection, only some users and targets can establish wireless connections with the UAV for data collection due to limited service coverage. Let $\mathcal{C}' \subseteq \mathcal{C}$ and $\mathcal{S}' \subseteq \mathcal{S}$ denote the subsets of users and targets of the UAV's service coverage, respectively. For users outside the set C', they can only upload data directly to the LEO satellite. For targets outside the set S', the UAV will not be able to collect data from them.

In this network, we assume that user c_n and target s_m are stationary ¹ on the ground while the UAV is flying at a constant altitude H [30]. Consequently, the positions of the user c_n , the target s_m , and the UAV are modeled using the two-dimensional Cartesian coordinate. Let $\mathbf{a}_n = [x_n, y_n]^T$, $\mathbf{b}_m = [x_m, y_m]^T$, and $\mathbf{q} = [x_u, y_u]^T$ denote the coordinates of user c_n , target s_m , and UAV, respectively. Note that the positions of all users and targets are known in advance to facilitate the design of the UAV trajectory. Subsequently, the data collection in multi-user multi-target scenarios is presented.

B. Data Collection in Multi-user Multi-target Scenarios

We adopt a path discretization approach to model the UAV's flight state [19], which requires fewer variables to represent the flight state compared to the conventional timeslot division method. As illustrated in Fig. 2, the UAV's flight distance is composed of K + 2 waypoints defined as

¹In this paper, we assume that users and targets are stationary, simplifying the system model. Dynamic scenarios may introduce some changes on the system, such as the mobility management, which involves the localization and movement prediction of users and targets. In addition, dynamic scenarios may need real-time optimization for UAVs in the system, posing higher demands on the problem formulation and algorithmic complexity. On the other hand, it should be noted that the system optimization in dynamic scenarios may involve the analysis and optimization in static scenarios of this work, which can serve as an important reference. In future work, we will investigate such dynamic scenarios involving unknown trajectories of users and targets. $\left\{\mathbf{q}_{k} = \left[x_{k}^{u}, y_{k}^{u}\right]^{T} \mid 0 \leq k \leq K+1, k \in \mathbf{Z}_{+}\right\}$, where \mathbf{q}_{k} is the coordinate of the k-th waypoint, and \mathbf{q}_{0} and \mathbf{q}_{K+1} denote the known initial and final waypoints of UAV, respectively. Among the K+2 waypoints, there are K+1 line segments, and we use $\boldsymbol{\Delta}_{k} = \mathbf{q}_{k+1} - \mathbf{q}_{k}$ to denote the k-th segment. Similar to the conventional time-slot division method, the UAV remains stationary at segment $\boldsymbol{\Delta}_{k}$, and it should meet the distance constraint, given by

$$\|\boldsymbol{\Delta}_k\| \le \min\left\{d_{max}, v_{max}t_k\right\},\tag{1}$$

where $\|\cdot\|$ returns the Euclidean distance, d_{max} represents the maximum distance in each segment, v_{max} is the UAV maximum speed, and t_k denotes the duration of segment Δ_k . For each segment, there are up to four schemes for data collection, which are presented as follows:

- Upload to UAV (UAV scheme): User c_n uploads its data to the UAV, and the data is stored in the UAV.
- Upload to LEO satellite (Direct scheme): User c_n directly uploads its data to the LEO satellite.
- Upload to LEO satellite (Relaying scheme): User c_n uploads its data to the LEO satellite with the help of the UAV.
- Sensed by UAV (Sensing scheme): The UAV actively sends perception signals to the sensing target s_m and acquires sensing data from the echoes.

Note that for each segment of the UAV trajectory, time division multiple access (TDMA) is employed among the N CUs and M sensing targets. Let $\beta_{n,k}$ and $\alpha_{m,k}$ denote the time allocated for the user c_n to upload data, and that allocated for the UAV to sense the target s_m at the segment Δ_k , respectively. Then, the time allocation constraint is given by

$$\sum_{m \in \mathcal{S}} \alpha_{m,k} + \sum_{g_n \in \mathcal{C}} \beta_{n,k} \le t_k.$$
(2)

Additionally, in one flight, the UAV trajectory \mathbf{Q} is comprised of the K + 2 waypoints, given by

$$\mathbf{Q} = [\mathbf{q}_0, \cdots, \mathbf{q}_k, \cdots, \mathbf{q}_{K+1}]. \tag{3}$$

At the segment Δ_k , the distances from the UAV to user c_n and that to target s_m are, respectively, written as

$$d_{n,k} = \sqrt{\left\|\mathbf{q}_k - \mathbf{a}_n\right\|^2 + H^2},\tag{4}$$

$$d_{m,k} = \sqrt{\|\mathbf{q}_k - \mathbf{b}_m\|^2 + H^2}.$$
 (5)

Note that the coordinate of segment Δ_k is defined as the coordinate of its initial position. In the next subsection, we detail the data collections of communication and sensing.

C. Data Collections of Communication and Sensing

Recall that there are up to four data collection schemes on the network in each segment, with the UAV and LEO satellite collecting data in TDMA mode, meaning that each user has only one time slot. For the first three data collection schemes, each user schedules up to one scheme to upload data. To simplify the description, we use three binary variables $x_{n,k}^{I}$, $x_{n,k}^{II}$, and $x_{n,k}^{III}$ to represent these three schemes. For example,



Fig. 2. Data collection in multi-user multi-target scenarios.

 $x_{n,k}^{I} = 1$ means that the UAV scheme is used for user c_n to upload data at segment Δk , or not otherwise. The same logic applies to the other two schemas. So there is a natural constraint, given by

$$x_{n,k}^{\mathbf{I}} + x_{n,k}^{\mathbf{II}} + x_{n,k}^{\mathbf{III}} \le 1, x_{n,k}^{j} \in \{0, 1\}, \ j \in \{\mathbf{I}, \mathbf{II}, \mathbf{III}\}.$$
(6)

Next, we will discuss in detail the data collection process through user upload and UAV sensing.

1) Upload to UAV (UAV scheme) with $x_{n,k}^{I} = 1$: In this scheme, the data is uploaded to the UAV by user c_n and stored at the UAV. Then, the amount of data collected is given by

$$R_{n,k}^{I} = \beta_{n,k} B_{I} \log_2 \left(1 + \frac{p_n \left| h_{n,k} \right|^2}{\sigma^2} \right), \tag{7}$$

where $B_{\rm I}$ represents the channel bandwidth of the c_n -UAV link, p_n is the transmit power of user c_n , σ^2 is the variance of the additive white Gaussian noise (AWGN), and $|h_{n,k}|^2$ represents the instantaneous channel gain between user c_n and the UAV at segment Δ_k . According to [31], we model the wireless channel as a LoS link, and the instantaneous channel gain is

$$|h_{n,k}|^2 = \frac{N_0}{d_{n,k}^2},$$
(8)

where $N_0 = \frac{G_t G_c \lambda^2}{(4\pi)^2}$ is the reference channel gain per unit distance for the CU-UAV link, in which G_t and G_c denote the transmit and receive antenna gains, respectively, and λ represents the wavelength.

2) Upload to LEO satellite (Direct scheme) with $x_{n,k}^{II} = 1$: In this scheme, user c_n directly uploads its data to the LEO satellite. For this scheme, the achievable upload data rate of user c_n at segment Δ_k is obtained as

$$r_{n,k}^{\rm II} = B_{\rm II} \log_2 \left(1 + f(c_1) p_n \right), \tag{9}$$

where $B_{\rm II}$ is the channel bandwidth of the c_n -LEO satellite link, and $f(c_1)$ represents the fading coefficient under channel condition c_1 [32]. Here, we adopt the three-state condition model [33], where f(c) equals 1.0, 3.46, and 5.03, corresponding to the bad, medium, and good conditions depending on the weather, respectively. In this scheme, the amount of data collected is $R_{n,k}^{\rm II} = \beta_{n,k} r_{n,k}^{\rm II}$. 3) Upload to LEO satellite (Relaying scheme) with $x_{n,k}^{III} = 1$: In this scheme, user c_n uploads its data to the LEO satellite with the help of the UAV, where the UAV serves as a halfduplex decode-and-forward (DF) relay to forward the user's data to the LEO satellite. For this scheme, the achievable upload data rate between user c_n and the LEO satellite at segment Δ_k is

$$r_{n,k}^{\text{III}} = \frac{1}{2} \min\left(B_1 \log_2\left(1 + \frac{p_n |h_{n,k}|^2}{\sigma^2}\right), r_u\right), \quad (10)$$

with

$$r_u = B_{\text{III}} \log_2 \left(1 + f(c_2) p_u \right),$$
 (11)

where r_u is the achievable rate of the UAV-LEO satellite link, $B_{\rm III}$ and c_2 are the corresponding channel bandwidth of the satellite and channel condition of the UAV-LEO satellite link, and $\frac{1}{2}$ comes from the two-phase data transmission. Although the upload data rate of this scheme depends on the minimum upload rate of the two-hop link, in practice, the first-hop link tends to be worse than the second-hop link in terms of upload data rate. This often happens as the satellite communication can utilize the microwave band, which has a wide available bandwidth, thus $B_{\rm III}$ is much larger than $B_{\rm I}$ [12], [34]. Therefore, the upload data rate under this scheme is mainly limited by the first hop, i.e., $r_{n,k}^{\rm III} = \frac{1}{2}B_{\rm I}\log_2\left(1 + \frac{p_n|h_{n,k}|^2}{\sigma^2}\right)$, leading to that the amount of data collected is $R_{n,k}^{\rm III} = \beta_{n,k}r_{n,k}^{\rm III}$.

Comparing UAV scheme and Relaying scheme, we can see that for this collaborative data collection network, if a user is within the set C' and meanwhile the UAV has enough memory to store the data, then UAV scheme tends to be used for data collection because the data rate of UAV scheme is about twice that of Relaying scheme. On the contrary, when the amount of data stored in the UAV reaches its capacity limit, the Relaying scheme tends to be used for data collection.

4) Sensed by UAV (Sensing scheme): In this scheme, the UAV actively sends perception signals to the sensing target and acquires sensing data from the echoes. In this way, the radar estimation information rate is used to measure the sensing performance [24], [27], [28]. According to [24], the amount of data sensed from target s_m at segment Δ_k is expressed as

$$R_{m,k}^{s} = \frac{1}{2} \alpha_{m,k} B_{\rm I} \log_2 \left(1 + \frac{2B_{\rm I} \mu \gamma^2 B_{\rm I}^2 \sigma_{pre}^2 p_s \left| h_{m,k}^s \right|^2}{\sigma^2} \right)_{(12)}^{\frac{\sigma}{\mu B_{\rm I}}}$$

where p_s is the UAV sensing power, μ is the radar pulse duration, γ is a constant determined by the shape of the radar waveform's power spectral density, σ_{pre}^2 is the variance of the predicted radar return, and δ is the radar duty factor. Besides, $\left|h_{m,k}^s\right|^2$ is the channel gain of the radar detection link from target s_m to the UAV at segment Δ_k , given by

$$\left|h_{m,k}^{s}\right|^{2} = \frac{N_{1}}{d_{m,k}^{4}},\tag{13}$$

where $N_1 = \frac{G_t G_r \lambda^2 \epsilon}{(4\pi)^3}$ is the reference channel gain per unit distance of the UAV-target-UAV link similar to (8), in which

 G_r is the UAV radar receiver antenna gain and ϵ is the radar and thus cross-section [21].

III. PROBLEM FORMULATION

In this section, we elaborate on the optimization problem, along with relevant constraints, aimed at maximizing the total amount of data collected.

A. Constraint Analysis

First, we detail the previously mentioned subsets \mathcal{C}' and \mathcal{S}' based on the received SNR. Whether a user or target belongs to the \mathcal{C}' or \mathcal{S}' subset depends on whether the corresponding received signal-to-noise ratio satisfies a predefined threshold. Mathematically, the received communication SNR $\Phi_{n,k}^c$, from the user $c_n \in \mathcal{C}'$ to the UAV at segment Δ_k , should satisfy

$$\Phi_{n,k}^{c} = \frac{p_n \left| h_{n,k} \right|^2}{\sigma^2} \ge \Phi_{th}^{c}, \tag{14}$$

where Φ_{th}^c is the predetermined communication SNR threshold. Equivalently, (14) can be expressed as

$$\|\mathbf{q}_k - \mathbf{a}_n\|^2 \le \frac{N_0 p_n}{\Phi_{th}^c \sigma^2} - H^2.$$
 (15)

Thus, at segment Δ_k , subset C' can be given by

$$\mathcal{C}_{k}^{\prime} = \left\{ g_{n} \in \mathcal{C} \mid \left\| \mathbf{q}_{k} - \mathbf{a}_{n} \right\|^{2} \leq \frac{N_{0} p_{n}}{\Phi_{th}^{c} \sigma^{2}} - H^{2} \right\}.$$
(16)

To facilitate the description of all segments, we denote the above relationship as a binary variable $c_{n,k}$, given by

$$c_{n,k} = \begin{cases} 1, & \text{If } \|\mathbf{q}_k - \mathbf{a}_n\|^2 \le \frac{N_0 p_n}{\Phi_{th}^c \sigma^2} - H^2 \\ 0, & \text{Else} \end{cases}$$
(17)

where $c_{n,k} = 1$ indicates that user c_n can upload its data by the assistance of the UAV at segment Δ_k , while $c_{n,k} = 0$ indicates that the UAV cannot serve user c_n . In sight of the above, (6) can be rewritten as

$$\begin{cases} x_{n,k}^{\mathrm{I}} + x_{n,k}^{\mathrm{II}} + x_{n,k}^{\mathrm{III}} = 1, & \text{If } c_{n,k} = 1\\ x_{n,k}^{\mathrm{I}} = 0, x_{n,k}^{\mathrm{II}} = 1, x_{n,k}^{\mathrm{III}} = 0, & \text{Else} \\ x_{n,k}^{j} \in \{0,1\}, \ j \in \{\mathrm{I}, \mathrm{II}, \mathrm{III}\}. \end{cases}$$
(18)

Similar to (14), the sensing SNR $\Phi_{m,k}^s$, from the UAV to target $s_m \in S'$ and back to the UAV at segment Δ_k , should satisfy

$$\Phi_{m,k}^{s} = \frac{N_2 \left| h_{m,k}^{s} \right|^2}{\sigma^2} \ge \Phi_{th}^{s}, \tag{19}$$

with

$$N_2 = 2B_{\rm I}\mu\gamma^2 B_{\rm I}^2\sigma_{pre}^2 p_s,\tag{20}$$

where Φ_{th}^s is the predetermined sensing SNR threshold. And (19) is equivalent to

$$\left\|\mathbf{q}_{k}-\mathbf{b}_{m}\right\|^{2} \leq \sqrt{\frac{N_{2}N_{1}}{\Phi_{th}^{s}\sigma^{2}}}-H^{2},$$
(21)

and thus, the subset \mathcal{S}'_k is given by

$$\mathcal{S}_{k}^{\prime} = \left\{ s_{m} \in \mathcal{S} \mid \left\| \mathbf{q}_{k} - \mathbf{b}_{m} \right\|^{2} \leq \sqrt{\frac{N_{2}N_{1}}{\Phi_{th}^{s}\sigma^{2}}} - H^{2} \right\}.$$
 (22)

Similar to (17), we use a binary variable $s_{m,k}$ to represent the above relationship of all segments, given by

$$s_{m,k} = \begin{cases} 1, & \text{If } \|\mathbf{q}_k - \mathbf{b}_m\|^2 \le \sqrt{\frac{N_2 N_1 p_s}{\Phi_{th}^s \sigma^2}} - H^2 \\ 0, & \text{Else} \end{cases}$$
(23)

where $s_{m,k} = 1$ indicates that the UAV is able to sense data from s_m at segment Δ_k , or fails otherwise.

Second, in order to ensure fairness among users, i.e., each user should achieve a minimum upload data amount Q_n , denoted as (24), which is at the top of the next page. Moreover, in practice, the radar estimation information rate is normally lower than the upload data rate [24]. To ensure that a sufficient amount of data sensed is collected, the amount of data sensed per target needs to satisfy a constraint, which is given by

$$\sum_{k=0}^{K} s_{m,k} R_{m,k}^{s} \ge Q_{m}^{s},$$
(25)

where Q_m^s is the minimum sensing data amount per target. Besides, there is a time limit for the whole data collection process. In other words, there is a maximum flight time limit for the UAV. Specifically, the UAV's prolonged flight not only depletes the energy it carries, leading to loss of control and safety accidents, but also increases the temperature of the UAV's fuselage, affecting the functioning of its key components, and in turn reducing flight stability. Thus, the time limit is given by

$$\sum_{k=0}^{K} t_k \le T_{th},\tag{26}$$

where T_{th} is the duration of data collection.

In particular, we should emphasize the limitations of the UAV in terms of power consumption and energy consumption and storage capacity. On the one hand, due to the finite battery capacity in practice, UAVs are not capable of continuous flight. Specifically, the energy consumption of UAVs mainly comes from three aspects: flight, communication, and sensing, of which flight is the largest contributor. Therefore, the other two aspects of energy consumption can be ignored [21], and the constraint on the total energy consumption of the UAV should follow the mathematical form given by

$$\sum_{k=0}^{K} t_k p_k^f \le E_{th},\tag{27}$$

where E_{th} is the threshold of the UAV energy consumption. p_k^f indicates the propulsion power for flight during segment Δ_k [19], given by

$$p_{k}^{f} = P_{0} \left(1 + \frac{3v_{k}^{2}}{U_{\text{tip}}^{2}} \right) + P_{s} \left(\sqrt{1 + \frac{v_{k}^{4}}{4v_{0}^{4}}} - \frac{v_{k}^{2}}{2v_{0}^{2}} \right)^{1/2} + \frac{1}{2} d_{0} \varpi_{0} s_{0} A_{0} v_{k}^{3},$$

$$(28)$$

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$$\sum_{k=0}^{K} \left(c_{n,k} \left(x_{n,k}^{\mathrm{I}} R_{n,k}^{\mathrm{I}} + x_{n,k}^{\mathrm{II}} R_{n,k}^{\mathrm{II}} + x_{n,k}^{\mathrm{III}} R_{n,k}^{\mathrm{III}} \right) + (1 - c_{n,k}) R_{n,k}^{\mathrm{II}} \right) \ge Q_n.$$
(24)

where $v_k \triangleq \frac{|\mathbf{q}k+1-\mathbf{q}k|}{t_k}$ represents the UAV speed at segment Δ_k . Notations P_0 and U_{tip} denote the blade profile power of hovering status and tip speed of the UAV rotor blade, respectively. Moreover, P_s and v_0 represent the blade induced power and mean rotor induced velocity in UAV hovering status, respectively. Additionally, d_0 , s_0 , ϖ_0 , and A_0 correspond to the fuselage drag ratio, rotor robustness, air density, and rotor disk area, respectively. On the other hand, it should be confirmed that the total amount of data uploaded by the users (when $x_{n,k}^I = 1$) and sensed from the targets will not exceed the memory capacity of the UAV. This can be done by satisfying the constraint, given by

$$Q_{u} = \sum_{k=0}^{K} \left(\sum_{n=1}^{N} c_{n,k} x_{n,k}^{I} R_{n,k}^{I} + \sum_{m=1}^{M} s_{m,k} R_{m,k}^{s} \right) \le Q_{u}^{th},$$
(29)

where Q_u^{th} is the threshold of the UAV memory capacity. The initial and final positions in a UAV flight trajectory are assumed to be known and are denoted by \mathbf{q}_I and \mathbf{q}_F , respectively, given by

$$\mathbf{q}_0 = \mathbf{q}_I, \ \mathbf{q}_{K+1} = \mathbf{q}_F. \tag{30}$$

Overall, all constraints in the collaborative SAG data collection network have been described, and the main notations are summarized in Table I for readability.

B. Problem Formulation

We aim to maximize the total amount of data collected in the SAG data collection network, denoted by Q_{sys} as follows,

$$Q_{\text{sys}} = \underbrace{\sum_{k=0}^{K} \sum_{n=1}^{N} c_{n,k} \left(x_{n,k}^{\text{I}} R_{n,k}^{\text{I}} + x_{n,k}^{\text{II}} R_{n,k}^{\text{II}} + x_{n,k}^{\text{III}} R_{n,k}^{\text{III}} \right)}_{Q_{1}}_{Q_{1}} + \underbrace{\sum_{k=0}^{K} \sum_{n=1}^{N} (1 - c_{n,k}) R_{n,k}^{\text{II}}}_{Q_{2}} + \underbrace{\sum_{k=0}^{K} \sum_{m=1}^{M} s_{m,k} R_{m,k}^{s}}_{Q_{3}}.$$
(31)

Note that Q_{sys} consists three parts. The first part Q_1 represents the total amount of data uploaded from users within the set C', the second part Q_2 represents the total amount of data uploaded from users out of set C', and the third part Q_3 represents the total amount of data sensed by the UAV.

Based on the previous analysis, we design a system optimization problem with the objective of maximizing the amount of data collected [35]–[37]. This involves optimizing the

TABLE I: Main Notations Summary

Notation	Definition
С, S	Set of communication users and sensing
	targets
$\mathcal{C}', \mathcal{S}'$	Set of users and targets within UAV service
	coverage
N, M	Number of communication users and sens-
	ing targets
c_n, s_m	<i>n</i> -th communication user and <i>m</i> -th sensing
	target
$\mathbf{\Delta}_k$	k-th segment
$\mathbf{a}_n, \mathbf{b}_m, \mathbf{q}_k$	Coordinate of user c_n , target s_m , and the
	UAV at segment Δ_k
Н	UAV flight altitude
K+2	Number of waypoints
d_{\max}	Maximum distance of each segment
v_{\max}	UAV maximum speed
t_k	Duration of segment Δ_k
$\beta_{n,k}, \alpha_{m,k}$	Time allocated variables
\mathbf{Q}	UAV trajectory
$d_{n,k}, d_{m,k}$	Distance from the UAV to user c_n and that
10,101 110,10	to target s_m at segment Δ_k
x^{I} , x^{II} , x^{III} ,	Scheme scheduling variables
$\begin{array}{cccc} n,\kappa & n,\kappa & n,\kappa \\ B^{\mathrm{I}} & B^{\mathrm{II}} & B^{\mathrm{II}} & B^{\mathrm{III}} \end{array}$	Amount of data collected from c_{-} at seq-
$n_{n,k}$, $n_{n,k}$, $n_{n,k}$	ment Λ_1 with different scheduling
Bi Bi Bi	Channel bandwidth
$p_{\rm m}$ $p_{\rm m}$ $p_{\rm m}$	Transmit power
h_{-1} , h^s .	Channel parameter of the UAV- c_{m} link and
$n_{n,k}, n_{m,k}$	the UAV-s _m link
σ^2	Noise power
$N_0 N_1$	Reference channel gain per unit distance for
1,0,1,1	the CU-UAV link and UAV-target-UAV link
f(c)	Fading coefficient of satellite link under
J(c)	channel condition c
$\Phi^c = \Phi^s$	Predetermined communication and sensing
$\mathbf{T}_{th}, \mathbf{T}_{th}$	SNR threshold
C 1 8 1	Indicator variable about sets C' and S'
$O_{m,k}, S_{m,k}$	Minimum unload data amount per user and
$\mathfrak{L}_n, \mathfrak{L}_m,$	minimum sensing data amount per target
O^{th} O	LIAV memory canacity and the total amount
Q_u , Q_{sys}	of data collected
T , F ,	Duration of data collection and threshold of
1 th, 12 th	the UAV energy consumption
g. g.	Initial and final positions in the UAV fight
$\mathbf{q}_{I},\mathbf{q}_{F}$	trajactory
	najectory

UAV trajectory, time allocation and data upload scheduling, formulated by

$$P0: \max_{\mathcal{V}} Q_{sys} \\
 s.t. (1), (2), (17), (18), (23), (32) \\
 (24), (25), (26), (27), (29), (30),$$

where $\mathcal{V} = \left\{ \mathbf{q}_k, t_k, c_{n,k}, s_{m,k}, \alpha_{m,k}, \beta_{n,k}, x_{n,k}^{\mathrm{I}}, x_{n,k}^{\mathrm{II}}, x_{n,k}^{\mathrm{III}}, x_{n,k}^{\mathrm{III}} \right\}$ is the set of the optimization variables, in which $0 \leq k \leq K+1$, $c_n \in \mathcal{C}$, and $s_m \in \mathcal{S}$.

After observing the optimization problem **P0**, we notice that there is coupling among the variables, resulting in a nonconvex problem. In addition, the optimization of the UAV trajectory increases the complexity of the problem, and the inclusion of discrete variables further complicates the analysis. As a result of these factors, the P0 problem evolves into a MINLP problem. Several approaches have been proposed to solve such problems, the most recent of which is the deep reinforcement learning (DRL) algorithm [38]. However, although the DRL algorithm shows unrivaled potential in handling such problems, the algorithm has several challenges in both the design phase and the training phase. Specifically, in the design phase of the algorithm, it faces complicated design problems regarding the huge state space and action space, while the sparse reward design problem is also a challenge. In addition, the design of the UAV trajectory also becomes a hurdle. In the training phase of the algorithm, it also will meet the problems of high training cost and unstable training [39]. Therefore, we develop an AIOA based on the BCD technique to solve the optimization problem **P0**, which will be detailed in the next section.

IV. PROPOSED METHOD

In this section, we first utilize the BCD technique to decompose the problem **P0**, resulting into two subproblems. Next, we analyze each subproblem individually, and based on these analyses, we proceed to design the AIOA framework to solve the problem **P0**.

A. Problem Decomposition

In this subsection, we decompose the original problem **P0** by using the BCD technique into two subproblems **P1** and **P2**, given by

$$P1: \max_{\mathcal{V}_1} Q_{\text{sys}} \\
 s.t. (1), (2), (17), (18), (23), (24), (33) \\
 (25), (26), (27), (29), (30),$$

where $V_1 = \{t_k, \mathbf{q}_k, c_{n,k}, s_{m,k}\}$, and

where $\mathcal{V}_2 = \left\{ \alpha_{m,k}, \beta_{n,k}, x_{n,k}^{\mathrm{II}}, x_{n,k}^{\mathrm{III}}, x_{n,k}^{\mathrm{III}} \right\}$. Note that subproblem **P1** focuses on optimizing the UAV trajectory with a given \mathcal{V}_2 and obtains a \mathcal{V}_1 . Then, the subproblem **P2** focuses on optimizing the data upload scheduling scheme and time allocation strategy with a given \mathcal{V}_1 and obtains a \mathcal{V}_2 . By alternately iterating these two subproblems, we can efficiently find a proper solution to the problem **P0**.

Next, we detail the solution process for P1 and P2.

B. Solutions

1) Solving **P1**: In **P1**, we observe that the variables $c_{n,k}$ and $s_{m,k}$ are not only binary but also coupled with \mathbf{q}_k , while the objective function and constraints in (24), (25), (27), and (29) are non-convex, making it very difficult to solve. To address **P1**, we first employ the big-M method to reconstruct the non-convex parts [40], and then solve the non-convex constraints by the SCA algorithm.

We start to solve P1 with restructuring the constraints in (17) and (23). Specifically, to use big-M method with the

constraints in (17) and (23), we introduce two slack variables $\xi_{1,n,k}$ and $\xi_{2,m,k}$, such that

$$\xi_{1,n,k} \ge \|\mathbf{q}_k - \mathbf{a}_n\|^2, \ \xi_{2,m,k} \ge \|\mathbf{q}_k - \mathbf{b}_m\|^2,$$
 (35)

and the constraints in (17) and (23) are converted to

$$c_{n,k} = \begin{cases} 1, & \text{If } \xi_{1,n,k} \le \frac{N_0 p_n}{\Phi_{th}^c \sigma^2} - H^2 \\ 0, & \text{Else} \end{cases}$$
(36)

and

$$s_{m,k} = \begin{cases} 1, & \text{If } \xi_{2,m,k} \le \sqrt{\frac{N_2 N_1}{\Phi_{th}^s \sigma^2}} - H^2 \\ 0. & \text{Else} \end{cases}$$
(37)

Then, according to the big-M method, the constraints in (36) and (37) can be rewritten as

$$\begin{cases} \xi_{1,n,k} - \mathbb{M}(1 - c_{n,k}) \le N_3 < \xi_{1,n,k} + \mathbb{M}c_{n,k}, \\ \xi_{2,m,k} - \mathbb{M}(1 - s_{m,k}) \le N_4 < \xi_{2,m,k} + \mathbb{M}s_{m,k}, \\ c_{n,k} \in \{0,1\}, \ s_{m,k} \in \{0,1\}, \end{cases}$$
(38)

where

$$N_3 = \frac{N_0 p_n}{\Phi_{th}^c \sigma^2} - H^2, \ N_4 = \sqrt{\frac{N_1 N_2}{\Phi_{th}^s \sigma^2}} - H^2,$$
(39)

and \mathbb{M} is a sufficiently large constant.

Second, based on the converted constraints in (35) and (38), we can restructure and handle the non-convex functions in **P1**. Specifically, for the non-convex UAV capacity constraint in (29), we introduce another two slack variables $\xi_{3,n,k}$ and $\xi_{4,m,k}$, such that

$$\xi_{3,n,k} \ge x_{n,k}^{\mathrm{I}} R_{n,k}^{\mathrm{I}} = x_{n,k}^{\mathrm{I}} \beta_{n,k} B_1 \log_2 \left(1 + \frac{p_n N_0}{\sigma^2 \left(\left\| \mathbf{q}_k - \mathbf{a}_n \right\|^2 + H^2 \right)} \right), \quad (40)$$

and

$$\xi_{4,m,k} \ge R_{m,k}^{s} = \alpha_{m,k} \frac{\delta}{2\mu} \log_2 \left(1 + \frac{N_2 N_1}{\sigma^2 \left(\|\mathbf{q}_k - \mathbf{b}_m\|^2 + H^2 \right)^2} \right).$$
(41)

Then, the constraint in (29) can be rewritten as

$$\sum_{k=0}^{K} \left(\sum_{n=1}^{N} c_{n,k} \xi_{3,n,k} + \sum_{m=1}^{M} s_{m,k} \xi_{4,m,k} \right) \le Q_u^{th}.$$
(42)

By employing the big-M method, (42) can be rewritten as

$$\begin{cases} \sum_{k=0}^{K} \left(\sum_{n=1}^{N} \xi_{5,n,k} + \sum_{m=1}^{M} \xi_{6,m,k} \right) \leq Q_{u}^{th}, \\ \xi_{3,n,k} - \mathbb{M} \left(1 - c_{n,k} \right) \leq \xi_{5,n,k} \leq \xi_{3,n,k} + \mathbb{M} \left(1 - c_{n,k} \right), \\ \xi_{4,m,k} - \mathbb{M} \left(1 - s_{m,k} \right) \leq \xi_{6,m,k} \leq \xi_{4,m,k} + \mathbb{M} \left(1 - s_{m,k} \right), \\ - \mathbb{M}c_{n,k} \leq \xi_{5,n,k} \leq \mathbb{M}c_{n,k}, \\ - \mathbb{M}s_{m,k} \leq \xi_{6,m,k} \leq \mathbb{M}s_{m,k}, \end{cases}$$

$$(43)$$

where $\xi_{5,n,k}$ and $\xi_{6,m,k}$ are auxiliary variables. However, the constraint in (29) is still non-convex due to newly introduced

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constraints in (40) and (41). To handle this issue, we introduce a slack variables $\xi_{7,n,k}$ and $\xi_{8,m,k}$ to (40) and (41), such that

$$\xi_{7,n,k} \le \|\mathbf{q}_k - \mathbf{a}_n\|^2 + H^2,$$
 (44)

$$\xi_{8,m,k} \le \left(\|\mathbf{q}_k - \mathbf{b}_m\|^2 + H^2 \right)^2,$$
 (45)

and the constraints in (40) and (41) can be rewritten as

$$\xi_{3,n,k} \ge x_{n,k}^{\mathrm{I}} \beta_{n,k} B_1 \log_2 \left(1 + \frac{p_n N_0}{\sigma^2 \xi_{7,n,k}} \right), \quad (46)$$

$$\xi_{4,m,k} \ge \alpha_{m,k} \frac{\delta}{2\mu} \log_2 \left(1 + \frac{N_2 N_1}{\sigma^2 \xi_{8,m,k}} \right). \quad (47)$$

It is not difficult to find that the constraint in (29) has been decoupled and the constraints except (44) and (45) are convex. While, in the non-convex constraints (44) and (45), the right-hand-side (RHS) of each constraint is a convex function with respect to \mathbf{q}_k , and the global concave lower bound of the RHS can be obtained based on any given localized point \mathbf{q}_k^i in the *i*-th iteration by the SCA algorithm. Specifically, by using the fact that first-order Taylor expansion is a global lower bound of a convex function, the lower bounds of the RHS of (44) and (45) can be denoted as

$$\|\mathbf{q}_{k} - \mathbf{a}_{n}\|^{2} + H^{2} \ge \|\mathbf{q}_{k}^{i} - \mathbf{a}_{n}\|^{2} + 2\left(\mathbf{q}_{k}^{i} - \mathbf{a}_{n}\right)^{T}\left(\mathbf{q}_{k} - \mathbf{q}_{k}^{i}\right) + H^{2} \ge \xi_{7,n,k},$$
(48)

and

$$\left(\left\| \mathbf{q}_{k} - \mathbf{b}_{m} \right\|^{2} + H^{2} \right)^{2} \geq \left(\left\| \mathbf{q}_{k}^{i} - \mathbf{b}_{m} \right\|^{2} + H^{2} \right)^{2} + 4 \left(\left\| \mathbf{q}_{k}^{i} - \mathbf{b}_{m} \right\|^{2} + H^{2} \right) \left(\mathbf{q}_{k}^{i} - \mathbf{b}_{m} \right)^{T} \left(\mathbf{q}_{k} - \mathbf{q}_{k}^{i} \right) \geq \xi_{8,m,k}.$$

$$(49)$$

For the non-convex constraints in (24) and (25) and nonconcave objective function, we use the same method mentioned above. Specifically, we introduce slack variables, such as

$$\tau_{1,n,k} \leq x_{n,k}^{\mathrm{I}} R_{n,k}^{\mathrm{I}} + x_{n,k}^{\mathrm{II}} R_{n,k}^{\mathrm{II}} + x_{n,k}^{\mathrm{III}} R_{n,k}^{\mathrm{III}}$$

$$= \left(x_{n,k}^{\mathrm{I}} + \frac{1}{2} x_{n,k}^{\mathrm{III}} \right) \beta_{n,k} B_{1} \times$$

$$\log_{2} \left(1 + \frac{p_{n} N_{0}}{\sigma^{2} \left(\|\mathbf{q}_{k} - \mathbf{a}_{n}\|^{2} + H^{2} \right)} \right)$$

$$+ x_{n,k}^{\mathrm{II}} R_{n,k}^{\mathrm{II}} \triangleq R_{n,k},$$
(50)

and

$$\tau_{2,m,k} \leq R_{m,k}^{s}$$
$$= \frac{1}{2} \alpha_{m,k} \frac{\delta}{\mu} \log_2 \left(1 + \frac{N_2 N_1}{\sigma^2 \left(\left\| \mathbf{q}_k - \mathbf{b}_m \right\|^2 + H^2 \right)^2} \right).$$
(51)

With the above manipulations, the non-convex constraints in (24) and (25), and non-concave objective function of problem **P1** can be rewritten as

$$\sum_{k=0}^{K} \left(c_{n,k} \tau_{1,n,k} + (1 - c_{n,k}) R_{n,k}^{\mathrm{II}} \right) \ge Q_n, \qquad (52)$$

$$\sum_{k=0}^{K} s_{m,k} \tau_{2,m,k} \ge Q_m^s, \tag{53}$$

and

$$\sum_{k=0}^{K} \sum_{n=1}^{N} \left(c_{n,k} \tau_{1,n,k} + (1 - c_{n,k}) R_{n,k}^{\text{II}} \right) + \sum_{k=0}^{K} \sum_{m=1}^{M} s_{m,k} \tau_{2,m,k} \triangleq Q_{\text{sys}}^{\sim}.$$
(54)

Similar to (42), we can rewrite (52), (53) and \hat{Q}_{sys} by the big-M method as

$$\begin{cases} \sum_{k=0}^{K} \tau_{3,n,k} \ge Q_{n}, \\ \tau_{1,n,k} - \mathbb{M} \left(1 - c_{n,k} \right) \le \tau_{3,n,k} \le \tau_{1,n,k} + \mathbb{M} \left(1 - c_{n,k} \right), \\ R_{n,k}^{\mathrm{II}} - \mathbb{M} c_{n,k} \le \tau_{3,n,k} \le R_{n,k}^{\mathrm{II}} + \mathbb{M} c_{n,k}, \end{cases}$$
(55)

$$\begin{cases} \sum_{k=0}^{K} \tau_{4,m,k} \ge Q_{m}^{s}, \\ \tau_{2,m,k} - \mathbb{M} \left(1 - s_{m,k} \right) \le \tau_{4,m,k} \le \tau_{2,m,k} + \mathbb{M} \left(1 - s_{m,k} \right), \\ - \mathbb{M} s_{m,k} \le \tau_{4,m,k} \le \mathbb{M} s_{m,k}, \end{cases}$$
(56)

and

$$\sum_{k=0}^{K} \sum_{n=1}^{N} \tau_{3,n,k} + \sum_{k=0}^{K} \sum_{m=1}^{M} \tau_{4,m,k} \triangleq Q_{\text{sys}}^{lb}, \qquad (57)$$

where $\tau_{3,n,k}$ and $\tau_{4,m,k}$ are auxiliary variables. Although we have achieved the decoupling of variables by reconstructing (24), (25) and the objective function through the big-M method, the newly introduced constraints in (50) and (51) are still non-convex. We adopt the SCA algorithm to address this issue. Specifically, similar to (48) and (49), the lower bounds are obtained by using the first-order Taylor expansion [19], and then (50) and (51) can be rewritten as

$$R_{n,k} \geq \left(x_{n,k}^{\mathrm{I}} + \frac{1}{2}x_{n,k}^{\mathrm{III}}\right)\beta_{n,k}B_{\mathrm{I}}\times \left(N_{5} - N_{6}\left(\|\mathbf{q}_{k} - \mathbf{a}_{n}\|^{2} - \|\mathbf{q}_{k}^{i} - \mathbf{a}_{n}\|^{2}\right)\right) + x_{n,k}^{\mathrm{II}}R_{n,k}^{\mathrm{II}} \geq \tau_{1,n,k},$$

$$(58)$$

and

$$R_{m,k}^{s} \ge \alpha_{m,k} \frac{\delta}{2\mu} \times \left(N_{7} - N_{8} \left(\left\| \mathbf{q}_{k} - \mathbf{b}_{n} \right\|^{2} - \left\| \mathbf{q}_{k}^{i} - \mathbf{b}_{n} \right\|^{2} \right) \right) \ge \tau_{2,m,k},$$
(59)

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$$\begin{cases} N_{5} = \log_{2} \left(1 + \frac{p_{n}N_{0}/\sigma^{2}}{\|\mathbf{q}_{k}^{i} - \mathbf{a}_{n}\|^{2} + H^{2}} \right), \\ N_{6} = \frac{p_{n}N_{0}/\sigma^{2}/\left(\|\mathbf{q}_{k}^{i} - \mathbf{a}_{n}\|^{2} + H^{2}\right)^{2}}{\ln 2\left(1 + \frac{p_{n}N_{0}/\sigma^{2}}{\|\mathbf{q}_{k}^{i} - \mathbf{a}_{n}\|^{2} + H^{2}} \right)}, \\ N_{7} = \log_{2} \left(1 + \frac{N_{2}N_{1}/\sigma^{2}}{\left(\|\mathbf{q}_{k}^{i} - \mathbf{b}_{n}\|^{2} + H^{2}\right)^{2}} \right), \\ N_{8} = \frac{2N_{2}N_{1}/\sigma^{2}/\left(\ln 2\left(\|\mathbf{q}_{k}^{i} - \mathbf{b}_{n}\|^{2} + H^{2}\right)^{2}\right)}{\left(N_{2}N_{1}/\sigma^{2} + \left(\|\mathbf{q}_{k}^{i} - \mathbf{b}_{n}\|^{2} + H^{2}\right)^{2}\right)}. \end{cases}$$
(60)

Therefore, the variables in the constraints (24), (25), (29) and objective function have been decoupled, and the derived non-convex constraints have been solved by the SCA algorithm. However, the problem P1 remains intractable because of the existence of the UAV energy constraint in (27). Fortunately, an effective solution to this problem has been proposed in [19]. Specifically, a relaxed upper bound has been used in the second term of (27), such as

$$G_{k} \geq t_{k} \left(\sqrt{1 + \frac{v_{k}^{4}}{4v_{0}^{4}}} - \frac{v_{k}^{2}}{2v_{0}^{2}} \right)^{1/2}$$

$$= \sqrt{\sqrt{t_{k}^{4} + \frac{\|\mathbf{q}_{k+1} - \mathbf{q}_{k}\|^{4}}{4v_{0}^{4}}} - \frac{\|\mathbf{q}_{k+1} - \mathbf{q}_{k}\|^{2}}{2v_{0}^{2}},$$
(61)

where G_k is a slack variable, and the constraint in (27) is rewritten as

$$\sum_{k=0}^{K} P_0 t_k \left(1 + \frac{3v_k^2}{U_{\rm tip}^2} \right) + P_s G_k + \frac{1}{2} d_0 \varpi_0 s_0 A_0 t_k v_k^3 \le E_{th}.$$
(62)

While, (61) is equivalent to

$$G_k^2 + \frac{\|\mathbf{q}_{k+1} - \mathbf{q}_k\|^2}{v_0^2} \ge \frac{t_k^4}{G_k^2}.$$
 (63)

For the non-convex constraint in (63), the left-hand-side (LHS) is a convex function with respect to G_k and \mathbf{q}_k , and RHS is a convex function with respect to t_k and G_k . Thus, the global lower bound of the LHS in (63) can be given by the first-order Taylor expansion, which is

$$G_{k}^{2} + \frac{\left\|\mathbf{q}_{k+1} - \mathbf{q}_{k}\right\|^{2}}{v_{0}^{2}} \geq \left(G_{k}^{i}\right)^{2} + 2G_{k}^{i}\left(G_{k} - G_{k}^{i}\right) \\ + \frac{1}{v_{0}^{2}}\left(2\left(\mathbf{q}_{k+1}^{i} - \mathbf{q}_{k}^{i}\right)^{T}\left(\mathbf{q}_{k+1} - \mathbf{q}_{k}\right) - \left\|\mathbf{q}_{k+1}^{i} - \mathbf{q}_{k}^{i}\right\|^{2}\right) \\ \geq \frac{t_{k}^{4}}{G_{k}^{2}},$$
(64)

where G_k^i is the given localized point in the *i*-th iteration.

Algorithm 1: The proposed UAV trajectory optimization algorithm.

1 Input \mathcal{V}_2 ;

- **2** Initialize iterative number i = 0, convergence accuracy θ_1 , and objective value $Q_{sys}^i = 0$;
- **3 Initialize** UAV trajectory $\{\mathbf{q}_k\}$ as $\{\mathbf{q}_k^{i,j}\}$;

4 Initialize $\{G_k\}$ as $\{G_k^i\}$;

- 5 repeat 6 i = i + 1;
- Solve the problem **P1-1** with the given V_2 , 7 $\{\mathbf{q}_k^{i-1}\}$, and $\{G_k^{i-1}\}$, and denote partial solutions as $\{\mathbf{q}_k^*\}$ and $\{G_k^*\}$;

8 Compute Q_{sys}^{lb} as $Q_{sys}^{lb,i}$; 9 Let $\{\mathbf{q}_{k}^{l}\} = \{\mathbf{q}_{k}^{k}\}$ and $\{G_{k}^{i}\} = \{G_{k}^{*}\};$ 10 until $|Q_{sys}^{lb,i} - Q_{sys}^{lb,i-1}| / Q_{sys}^{lb,i} \le \theta_{1};$

11 return $\{t_k, \mathbf{q}_k^i, c_{n,k}, s_{m,k}\}$

In the end, problem P1 can be rewritten as follows,

P1-1:

$$\max_{\mathcal{V}_{1-1}} Q_{sys}^{lb}$$
 (1), (2), (18), (26), (30), (35), (38),
 (65)

 s.t.
 (1), (2), (18), (26), (55) - (59), (62), (64),
 (65)

where $\mathcal{V}_{1-1} = \{t_k, \mathbf{q}_k, c_{n,k}, s_{m,k}, \xi_1 - \xi_8, \tau_1 - \tau_4, G_k\}.$

Overall, problem P1 has been transformed from a MINLP problem to a mixed-integer convex optimization problem P1-1 with binary variables $c_{n,k}$ and $s_{m,k}$. Fortunately, **P1-1** can be transformed into a convex optimization problem by traditional branch-and-bound or variable relaxation method, and then solved by a convex optimization solver such as MOSEK. Based on this, we propose a UAV trajectory optimization algorithm as shown in Algorithm 1, where G_k can be initialized by making the left and right hand sides of (61) equal according to the given \mathbf{q}_k and t_k . The above procedure is performed during each iteration until convergence.

2) Solving P2: Although problem P2 is a mixed integer programming problem involving binary variables $x_{n,k}^{I}$, $x_{n,k}^{II}$, and $x_{n,k}^{\text{III}}$, we can solve it by relaxing the integer variables to continuous real variables from 0 to 1. Thus, (18) can be converted to

$$\begin{cases} x_{n,k}^{\mathrm{I}} + x_{n,k}^{\mathrm{II}} + x_{n,k}^{\mathrm{III}} = 1, & \text{If } c_{n,k} = 1\\ x_{n,k}^{\mathrm{I}} = 0, x_{n,k}^{\mathrm{II}} = 1, x_{n,k}^{\mathrm{III}} = 0, & \text{Else} \end{cases}$$
(66)
$$x_{n,k}^{j} \in [0,1], \ j \in \{\text{I}, \text{II}, \text{III}\}.$$

Note that this relaxation removes the integer optimization, thus making the problem more tractable [41].

Then, P2 can be relaxed as follows:

$$\tilde{\mathbf{P2}}: \max_{\mathcal{V}_2} Q_{\text{sys}} \qquad (67)$$
s.t. (2), (24), (25), (29), (66).

However, $\tilde{P2}$ is still non-convex because of the high coupling among the variables. To solve this problem, we also use the Algorithm 2: The proposed algorithm for P2.

1 Input \mathcal{V}_1 ;

- 2 Initialize convergence accuracy θ_2 , \mathcal{V}_4 , Q_{sys}^1 and Q_{sys}^2 ; 3 repeat
- 4 Solve the problem **P2-1** with the given V_1 and V_4 , and obtain \mathcal{V}_3 ;
- Compute Q_{sys} as Q_{sys}^1 ; 5
- Solve the problem **P2-2** with the given V_1 and V_3 , 6 and obtain \mathcal{V}_4 ;

7 Compute Q_{sys} as Q_{sys}^2 ; 8 until $|Q_{\text{sys}}^2 - Q_{\text{sys}}^1| / Q_{\text{sys}}^2 \le \theta_2$; 9 return $\left\{ \alpha_{m,k}, \beta_{n,k}, x_{n,k}^I, x_{n,k}^{II}, x_{n,k}^{III} \right\}$

Algorithm 3: The proposed BCD based AIOA framework.

- 1 Initialize V_2 , iterative number I = 0, and maximum number of iterations I_{max} ;
- 2 while $I < I_{max}$ do
- Solve the problem **P1-1** with the given V_2 by 3 Algorithm 1;
- Solve the problem $\mathbf{P2}$ with the given \mathcal{V}_1 by 4 Algorithm 2;
- I = I + 1;5

6 end

7 return \mathcal{V}_1 , \mathcal{V}_2

BCD technique for $\tilde{P2}$. Specifically, the problem $\tilde{P2}$ can be decomposed into the subproblems P2-1 and P2-2, given by

P2-1:
$$\max_{\mathcal{V}_3} Q_{sys}$$

s.t. (24), (29), (66), (68)

where $\mathcal{V}_{3} = \left\{ x_{n,k}^{\text{I}}, x_{n,k}^{\text{II}}, x_{n,k}^{\text{III}} \right\}$, and **P2-2**: $\max_{\mathcal{V}_{4}} Q_{\text{sys}}$ s.t. (2), (24), (25), (29), (69)

where $\mathcal{V}_4 = \{\alpha_{m,k}, \beta_{n,k}\}$. The problems **P2-1** and **P2-2** are linear optimization problems with respect to the corresponding variables, which can be solved easily by means of several convex tools, such as CVX. By alternately solving the problems P2-1 and P2-2, we can obtain a time allocation strategy and a data upload scheduling scheme. The details are shown in Algorithm 2.

C. Design of AIOA based on BCD technique

Now, we present the AIOA framework to solve the problem P0 with the objective of obtaining a locally optimal solution for \mathcal{V} . The specific steps of the framework are detailed in Algorithm 3. Specifically, in step 3 of the Algorithm 3, we solve **P1-1** with the given V_2 in each iteration while updating the local points, until the algorithm converges to finally return \mathcal{V}_1 . Similarly, in step 4 of algorithm 3, we aim to solve **P2** with the given \mathcal{V}_1 , in which we solve **P2-1** and **P2-2** alternately

until the algorithm converges and \mathcal{V}_2 is finally obtained. It is worth noting that the convergence of AIOA can be guaranteed by varying the initial points, and its convergence analysis is similar to the work in [42]–[44]. Additionally, unlike existing methods, our proposed AIOA framework deals with complicated nonconvex optimization problems mainly through AO and SCA. This approach leads to an optimized solution in each subproblem, eventually yielding an appropriate solution to the original problem, which is often applicable to problems with a well-defined structure.

The computational complexity of Algorithm 3 can be analyzed as follows. As Algorithm 3 is composed of Algorithms 1 and 2, we first analyze the computational complexity of Algorithms 1 and 2. For Algorithm 1, its computational complexity is $\mathcal{O}\left(J_1\left((K+2)\left(7\left(M+N\right)+3\right)\right)^{3.5}\right)$, where $(K+2)\left(7\left(M+N\right)+3\right)$ represents the number of variables and J_1 is the number of iterations required [43]. For Algorithm 2, its computational complexity mainly lies in steps 4 and 6. Specifically, the complexity of step 4 is $\mathcal{O}\left(\left(3\left(K+2\right)N\right)^{3.5}\right)$, and the complexity of step 6 is $\mathcal{O}\left(\left((K+2)\left(M+N\right)\right)^{3.5}\right)$. Thus, the computational complexity of Algorithm 2 is $\mathcal{O}\left(J_2\left(\left(3\left(K+2\right)N\right)^{3.5}+\left(\left(K+2\right)\left(M+N\right)\right)^{3.5}\right)\right)$, where J_2 denotes the number of iterations required for convergence in Algorithm 2 [43]. In summary, the computational complexity of Algorithm 3 can be given by

$$\mathcal{O}\left(I_{\max}\left(J_1\left((K+2)\left(7\left(M+N\right)+3\right)\right)^{3.5}+J_2\left((3N(K+2))^{3.5}+\left((K+2)\left(M+N\right)\right)^{3.5}\right)\right)\right).$$
(70)

V. SIMULATION RESULTS AND DISCUSSIONS

A. Parameter Settings

This part presents some general simulation parameter settings. If not specified, the number of communication users and sensing targets are both set to two, where the locations of users are $[20, 80]^T$ m and $[80, 80]^T$ m, while the locations of targets are $[40, 30]^T$ m and $[60, 30]^T$ m. The UAV's initial and final locations are $\mathbf{q}_I = [0, 50]^T$ m and $\mathbf{q}_F = [100, 50]^T$ m. The UAV flight altitude H is 100 m, and its maximum speed is 30 m/s. Moreover, the wireless bandwidth $B_{\rm I}$ and $B_{\rm II}$ are set to 1 MHz and 10 MHz, while the transmit power at each user is set to 0.1 W, with the noise variance σ^2 set to -110 dBm. The channel gain per unit distance for the CU-UAV and UAV-target-UAV links is -60 dB. In further, for the parameters related to the radar estimation information rate in (12), we follow the settings in [27], where the UAV sensing power p_s is 10 W. Meanwhile, for the parameters related to the UAV propulsion power in (28), we follow the settings in [19]. Furthermore, the minimum amount of upload data per user Q_n is 10 Mbits, while the minimum amount of sensing data per target Q_m^s is 10 Kbits. The UAV memory capacity Q_u^{th} is 100 Mbits, the duration of data collection T_{th} is 30 s, and the This article has been accepted for publication in IEEE Journal on Selected Areas in Communications. This is the author's version which has not been fully edited and content may change prior to final publication. Citation information: DOI 10.1109/JSAC.2024.3459079



Fig. 3. Convergence of AIOA and internal algorithms under different settings.

TABLE II: Main Parameter Settings

Parameters	Value
UAV flight altitude H	100 m
UAV maximum speed v_{max}	30 m/s
Wireless bandwidth $B_{\rm I}$ and $B_{\rm II}$	1 MHz, 10 MHz
Transmit power at users	0.1 W
Noise variance σ^2	-110 dBm
Average channel gain N_0 and N_1	-60 dB
UAV sensing power p_s	10 W
Minimum amount of upload data per user Q_n	10 Mbits
Minimum amount of sensing data per target Q_m^s	10 Kbits
UAV memory capacity Q_u^{th}	100 Mbits
Duration of data collection T_{th}	30 s
Energy consumption threshold E_{th}	10 kJ

energy consumption threshold E_{th} is 10 kJ, respectively. In the end, the carrier frequency of the CU-UAV link and UAV-target-UAV link is C-band with 5 GHz, while the carrier frequency of CU-LEO satellite link and UAV-LEO satellite link is Ka-band with 35 GHz. For convenience, the main parameter settings are summarized in Table II.

B. Simulation Results

In this subsection, we verify the effectiveness of the proposed AIOA using two criteria regarding UAV coverage. In addition, we provide several data collection schemes for comparison. For convenience, we use the following notations to denote the schemes:

- AIOA-I: The proposed AIOA framework with a wider UAV coverage optimizes the UAV trajectory, time allocation, and scheduling of data uploading, where $\sqrt{N_3} = \sqrt{N_4} = 65$ m.
- AIOA-II: The proposed AIOA framework with a narrow UAV coverage optimizes the UAV trajectory, time allocation, and scheduling of data uploading, where $\sqrt{N_3} = \sqrt{N_4} = 35$ m.
- All-LEO: All communication data are uploaded either directly to the LEO satellite (Direct scheme) or with the help of the UAV to the LEO satellite (Relaying scheme). In this scheme, except for variable $x_{n,k}^{I}$, all the other variables are optimized.

- SF: The UAV flies straight from the initial position q_I to the final position q_F at a constant speed, where the data uploading schedule is random and the time allocation strategy is optimized in this scheme.
- Non-UAV: In this scheme, we aim to present the effect of satellite communication in our considered scenario. Therefore, in this competing scheme, users share the time resource to upload communication data, and the sensing mission is ignored.

Note that the coverage of the UAV in the All-LEO and SF is the same as that of AIOA-I. These competing schemes are the simplified versions of our proposed AIOA with lower complexity, which can provide some insights for the performance of our scheme. Besides, two initial trajectories are considered,

- Initial trajectory 1: This initial trajectory represents that $\{\mathbf{q}_k^0\}_{k=0}^{K+1}$ is a simple straight-line path from \mathbf{q}_I to \mathbf{q}_F
- Initial trajectory 2: This initial trajectory is a complex M-shaped trajectory that starts from \mathbf{q}_I and passes through the points $[20, 80]^T$ m, $[50, 30]^T$ m and $[80, 80]^T$ m in sequence, finally reaching \mathbf{q}_F .

Fig. 3 shows the convergence of our proposed AIOA framework and the internal Algorithms 1 and 2. From Fig. 3, we can observe that the AIOA and Algorithms 1, 2 all converge in just a few iterations, which proves the fast convergence rate and effectiveness. Moreover, in Fig. 3(c), "initial trajectory 1" performs better than "initial trajectory 2" in the case of AIOA-I, which highlights the importance of choosing a suitable initial solution. Thus, if not specified, we then default to "initial trajectory 1". In further, AIOA-I performs better than AIOA-II. Specifically, after several iterations, AIOA-I can collect about 90 Mbits of data, compared to AIOA-II, which can only collect about 60 Mbits, about 50% improvement in the amount collected by AIOA-I. This is because the UAV can collect data at a wider range.

Figs. 4 and 5 show the UAV trajectories and the corresponding speed changes. We can observe through these figures that the UAV generally flies at its highest speed toward the users and targets. This strategy is chosen not only to obtain better channel conditions but also because, in the AIOA-II, the UAV should be close to the user and the target to help the users collect data and sense the targets. Particularly, it is evident from Figs. 4(a), 5(a), 4(b), and 5(b) that "Initial Trajectory 2"



(a) UAV trajectory of AIOA-I with initial trajectory 1.



(a) UAV speed variation of AIOA-I with initial trajectory 1.



(b) UAV trajectory of AIOA-I with initial trajectory 2.

Fig. 4. UAV trajectories under different settings.



(b) UAV speed variation of AIOA-I with initial trajectory 2.

Fig. 5. UAV speed under different settings.



(c) UAV trajectory of AIOA-II with initial trajectory 1.



(c) UAV speed variation of AIOA-II with initial trajectory 1.

is characterized by a longer duration of flight as opposed to hovering over either the users or the targets. Consequently, this results in "Initial Trajectory 2" collecting less data compared to "Initial Trajectory 1" for a given threshold T_{th} . Moreover, while the trajectories in Figs. 4(a) and 4(c) appear deceptively similar, but notable differences emerge when examining the speed variations in Figs. 5(a) and 5(c). Specifically, Fig. 5(a) illustrates four hover points, i.e., moments when the speed is 0, contrasted with six such points in Fig. 5(c). This discrepancy is attributable to the limited service coverage of the UAV in AIOA-II, necessitating a balance in positioning between the users and targets for optimal data upload and target sensing, thereby maximizing Q_{sys} . However, such positions often entail suboptimal channel conditions compared to closer proximity to either the users or targets, which leads to AIOA-II collecting less data than AIOA-I.

Fig. 6 shows the impact of the UAV maximum speed v_{max} on Q_{sys} , where the speed ranges from 10 m/s to 30 m/s. Observing Fig. 6, it is evident that increasing the UAV's maximum speed v_{max} enhances Q_{sys} in all schemes except for the SF and Non-UAV schemes. This improvement occurs because a higher maximum speed allows the UAV to quickly approach users and targets, meeting the conditions of subsets C' and S', and gaining better channel gains. Moreover, the AIOA-I consistently outperforms AIOA-II. Specifically, as the UAV maximum speed increases, Q_{sys} of the AIOA-I increases



Fig. 6. Total amount of data collected versus v_{max} .

roughly from 81 Mbits to 90 Mbits. In comparison, Q_{sys} of the AIOA-II only increases roughly from 57 Mbits to 60 Mbits. This is because, in the former, the UAV collects data farther from the users and targets. In addition, the proposed schemes, AIOA-I and AIOA-II outperform other data collection schemes by at least 38%, proving superior performance.

Fig. 7 shows the relationship between Q_{sys} and duration



Fig. 7. Total amount of data collected versus T_{th} , where v_{max} is 15 m/s.



Fig. 8. Data collection of our proposed AIOA-II under various data upload schemes and durations.

 T_{th} for different schemes, where T_{th} ranges from 30 s to 40 s and v_{max} is 15 m/s. Observing Fig. 7, we can see that with the increasing T_{th} , Q_{sys} with all schemes shows an increasing trend. In particular, the growth rates of Q_{sys} in the proposed schemes slow down when Q_{sys} reaches about 100 Mbits. Due to the fact that the UAV plays a core data collection role in the SAG network, and when Q_{sys} reaches the upper limit of 100 Mbits of UAV capacity, the UAV's role is limited, ultimately resulting in a slowdown in the growing trend of Q_{sys} . Moreover, the proposed schemes outperform other schemes for various T_{th} . Specifically, when $T_{th} = 40$ s, the Q_{sys} of All-LEO, SF, and Non-UAV schemes are approximately 92 Mbits, 100 Mbits, and 30 Mbits, respectively. In comparison, the Q_{sys} of AIOA-I and AIOA-II are 138 Mbits and 129 Mbits, respectively, showing the effectiveness of the AIOA scheme.

Fig. 8 illustrates the data collection capabilities of our proposed AIOA-II under various data upload schemes and durations T_{th} , where $T_{th} = \{30, 40\}$ s. To facilitate ob-



Fig. 9. Total amount of data collected versus Q_u^{th} , where v_{max} is 15 m/s.

servation, the Direct scheme in Fig. 8 combines the data uploaded both inside and outside the set C', with the amount of data within C' being zero. At the duration of $T_{th} = 30$ s, the UAV scheme collects data approximately 56 Mbits, the Direct scheme collects about 4 Mbits, the Relaying scheme gathers 0 Mbits, and the Sensing scheme accumulates 20 Kbits. When the duration increases to $T_{th} = 40$ s, the amount of data collected by the UAV scheme rises to roughly 100 Mbits of UAV capacity, the Direct scheme collects about 6 Mbits, the Relaying scheme gathers 24 Mbits, while the Sensing scheme remains at 20 Kbits. From these numerical comparisons, we observe that within this sensing-assisted SAG integrated network, the UAV scheme is the first choice for data upload, followed by the Relaying scheme, and finally the Direct scheme.

Fig. 9 shows the impact of memory capacity of the UAV Q_u^{th} on the total collected data Q_{sys} , where Q_u^{th} varies from 10 Mbits to 100 Mbits and v_{max} is 15 m/s. We can observe from Fig. 9 that except for the All-LEO and Non-UAV schemes, $Q_{\rm sys}$ first increases and then stabilizes with the increase of Q_u^{th} . The reason is that when Q_u^{th} is small, the UAV memory capacity dominates the effect on $Q_{\rm sys}$, and increasing Q_u^{th} can significantly increase Q_{sys} . However, when Q_u^{th} is large, the memory capacity constraint is negligible. Therefore, the performance gain of increasing Q_u^{th} on Q_{sys} becomes marginal. Moreover, the proposed AIOA outperforms other schemes with the increasing Q_u^{th} , which shows the effectiveness of the proposed AIOA. Besides, from Fig. 9, we perceive an interesting phenomenon that the Q_{sys} of the AIOA-II scheme is first lower and then higher than that of All-LEO when Q_u^{th} increases. This is due to the performance of the AIOA-II is limited by the service range, under a lower memory capacity Q_u^{th} . However, with a higher memory capacity Q_u^{th} , AIOA-II can take full advantage of these sufficient UAV memory resources to collect data, while the All-LEO scheme fails.

Fig. 10 shows the impact of Q_m^s on several aforementioned schemes' Q_{sys} , where Q_m^s ranges from 5 Kbits to 10 Kbits



Fig. 10. Total amount of data collected versus Q_m^s , where v_{max} is 15 m/s.

and v_{max} is 15 m/s. It can be observed from Fig. 10 that, as Q_m^s increases, Q_{sys} of all schemes shows a declining trend, except the Non-UAV scheme. This can be attributed to the fact that the radar estimation information rate is significantly lower than the upload data rate. As a result, the UAV needs to allocate more time resource to sense the targets. When T_{th} is maintained at a constant value, the extended sensing duration results in a decrease of $Q_{\rm sys}$. In particular, within the proposed schemes, there is an observable acceleration in the decrease rate of Q_{sys} when Q_m^s exceeds 8 Kbits, at which point Q_{sys} reaches the UAV memory capacity. This trend indicates that beyond the limitation of the UAV memory capacity, increasing demand for sensing amounts exerts a more pronounced impact on Q_{sys} . In addition, under different Q_m^s settings, the proposed schemes are better than the competing schemes, proving their effectiveness on data collection.

VI. CONCLUSION

In this paper, we considered a sensing-enabled integrated SAG data collection network in which the UAV could not only work alone to sense data from multiple targets, but also collaborate with the LEO satellite to collect communication data from multiple users. For this system, we first posed the optimization problem designed to maximize the total amount of data collected in the network while satisfying the constraints of UAV energy consumption, storage capacity, and minimum amount of sensor data per target. What's more, considering that the network included three layers and the UAV had dual functions of communication and sensing, this problem could be solved by jointly optimizing the scheduling of the users' data upload scheme, the UAV trajectory, and the allocation of communication and sensing time. We further designed the AIOA framework to find a feasible solution. Specifically, we alternately optimized the UAV trajectory, time allocation strategy, and data upload schedule in each iteration. Finally, simulation experiments validated the effectiveness of AIOA and its superiority over other benchmarks. In particular, the experimental results also showed that the AIOA was superior to the linear trajectories of the UAV in terms of the amount of data collected.

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